Wide Area Augmentation System Offline Monitoring Quarterly Report #2

1 April 2011 - 30 June 2011

21 October 2011

Prepared for:

Federal Aviation Administration

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Executive Summary

The WAAS Operations Team (AJW-192) was tasked with monitoring the Wide Area Augmentation System (WAAS) to ensure that the integrity requirements defined in Section 3.3 of Algorithm Contribution to HMI (A014-011). This report contains data collected and analyzed between 4/1/11 and 6/30/11. Data is collected from the WAAS Network and stored at the WAAS Support Facility (WSF) at the Mike Monroney Aeronautical Center in Oklahoma City, OK.

The primary evidence that WAAS meets the top level system integrity requirements relies on a mathematical proof supported by a comprehensive analysis of empirical data. The foundation of the proof is built upon a set of carefully constructed assertions. Some assertions require periodic monitoring to ensure that the physical environment has not changed or degraded in a manner that would invalidate the claim. Certain satellite failure modes which have a priori probabilities associated with them must be detected and corrected in a reasonable amount of time to limit the user's exposure to the failure.

The following assertions are monitored as called for in the Algorithm Contribution to HMI document:

- 1. Code-noise and Multipath
- 2. Code-carrier-incoherence (CCC)
- 3. Signal Quality Monitor (SQM)
- 4. Satellite Clock Runoff
- 5. Ionospheric Threats
- 6. Ephemeris Monitoring

Additional monitoring criteria have been added to the original list. These additional monitoring criteria include Wide-area Reference Station (WRS) antenna positions, L1L2 bias levels, missed WAAS User Messages, monitor trips, CNMP resets, accuracy, GEO CCC, and Space Weather. This report will also include major anomalies that occurred during the time period covered in this report.

Below is a summary of the criteria that were monitored for this report.

	T
Integrity Monitoring	
Code-carrier-coherence	5 trips (all on PRN 138)
(CCC)	
Code-noise and multipath	All data bounded
(CNMP)	
Signal-quality monitoring	None
SQM	None
34,11	
Satellite clock run-off	None
Ionospheric threat model	7 days of interest – April 2,
	3, 6, 11, 12, 20 and 30
Continuity Monitoring	
Continuity Monitoring	
Missed messages	67 on CRW, 27 on CRE and
	553 on AMR (unavailable
	data from 5/27 – 6/6)
External Monitoring	
Antenna Positioning	All sites within 5 cm.
Anomaly Investigations	
Santa Paula AMR SIS outage	~7 min SIS outage on June
	16th

Forward

The scope of this document is limited to analysis performed on data extracted from the WAAS system, or on data that would directly affect the WAAS system. Moreover, the target audience is the FAA WAAS management as well as the engineers that support the WAAS program. This includes (but is not necessarily limited to) federally employed personnel, contractors, sub-contractors, and other FAA WIPP support.

The data and information contained in this document is not for general use, as it may contain unexplained anomalies and/or data which may lead to unsupported conclusions. Any dissemination of this data should be coordinated with the appropriate management.

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1 Introduction

1.1 The definition of offline monitoring

The goal of Offline monitoring is to track the performance of WAAS, establish baseline performance and characterize anomalous behavior to determine if further investigation is necessary.

1.2 Elements of system monitoring

Monitoring addressed in this document can be categorized into five types, namely Integrity, Availability, Continuity, Accuracy and External Monitoring. Each category represents a class of performance that the system exhibits. The intent of this document is to provide a summary of results for several checks of each of the above types in conjunction with condensed plots that show at a glance quarterly performance. Each monitoring subsection contains a brief description of the relevant figures and tables along with a pointer to a section in Appendix A which contains more detailed (and more numerous) figures and tables.

1.2.1 Integrity

Integrity monitoring is viewed by many to be the most important type since as a breach of this class of performance represents a potentially hazardous situation. Loss of Integrity happens when an error in a user's position exceeds the protection limits that he computes. There are monitors in WAAS which internally ensure that these error bounds represent an over-bound of the generated errors. Each monitor has a slightly different method for ensuring integrity, and the individual monitor integrity methodology is described in their respective monitor subsections.

1.1.1 Availability

Availability Monitoring is straightforward, evaluates the coverage of WAAS over the time period in question. There are specifics to be defined for this type, namely the Alarm Limits (Vertical and Horizontal) as well as the coverage contour.

1.1.2 Continuity

Continuity monitoring refers to events which can cause a loss of availability but not a breach of integrity. Typically, this assessment looks at monitor trips, setting satellites unusable or any issue which would cause a loss of service.

1.1.3 Accuracy

Accuracy Monitoring refers to the ability of the WAAS corrections to provide an accurate estimate the user's position.

1.1.4 External Monitoring

External monitoring entails events external to the WAAS, including broadcasted ephemerides, plate-tectonic movement (antenna positions), space weather, etc., that can result in anomalous WAAS performance.

2 Integrity Monitoring

2.1 Code-noise and multipath (CNMP)

For Figures 2.1-1, 2.1-2 and 2.1-3, CNMP passes if the tails (red and blue lines) do not dip below zero on the vertical axis. If a dip below zero occurs inside of plus or minus 0.25 that event is not considered a failure. For Figure 2.1-4, if the values go above the marked threshold of 1, that event is a failure. No CNMP failures occurred during the second quarter of 2011.

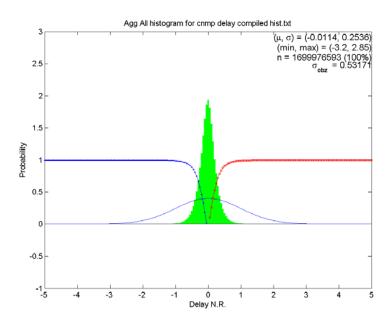


Figure 2.1-1 Aggregate CNMP Delay

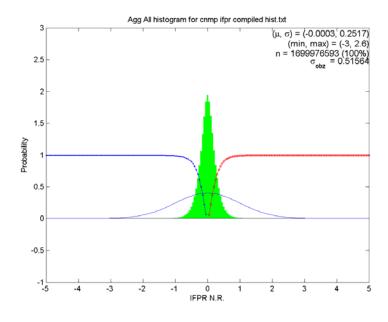


Figure 2.1-2 Aggregate CNMP IFPR

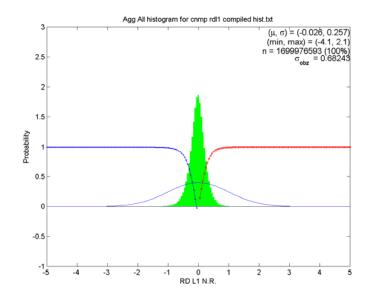


Figure 2.1-3 Aggregate CNMP RDL1

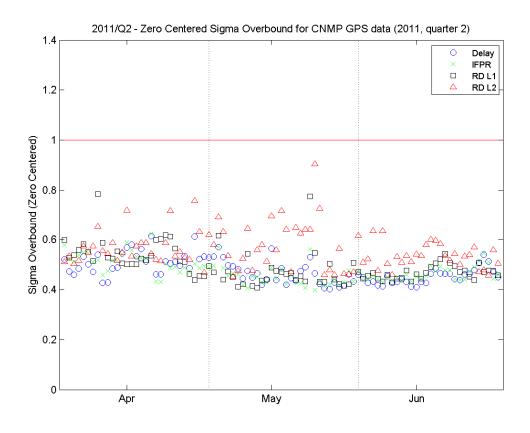


Figure 2.1-4 Daily GPS CNMP Aggregate zero-centered sigma overbound values

2.2 Code-carrier-coherence (CCC)

Anik, Galaxy 15 and all GPS satellites are monitored for CCC trips. All CCC monitor trips are investigated whenever a trip occurs to determine the cause. AMR is not currently monitored since it is not used as ranging source and its UDRE floor is 50m.

Data sources used in correlation and analysis:

- CCC test statistic
- UDRE threshold value
- CMCI measurements from NETS SQA
- WAAS Iono calculation
- L1/L5 Iono GUST calculation
- published planetary K_p and A_p values
- Chi² values

Table 1.1.4-2 Reported CCC Trips

CCC Trips for the 2nd quarter 2011											
UTC Time	PRN		C&V								
2011-06-13 10:07:35	138			ZTL							
2011-06-13 10:07:57	138	ZDC	ZLA	ZTL							
2011-06-13 10:12:46	138	ZDC									
2011-06-18 16:05:34	138	ZDC	ZLA	ZTL							
2011-06-18 16:05:53	138	ZDC		ZTL							

CCC Trips on CRE/138 are ongoing and under investigation.

2.3 CCC aggregate plots

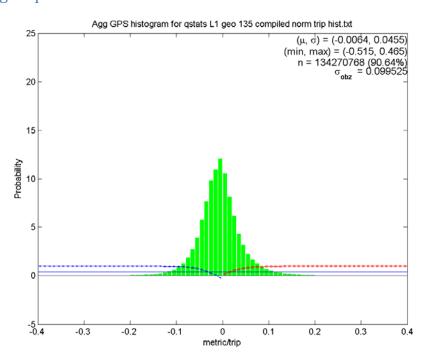


Figure 2.3-1 CCC GPS aggregate for L1

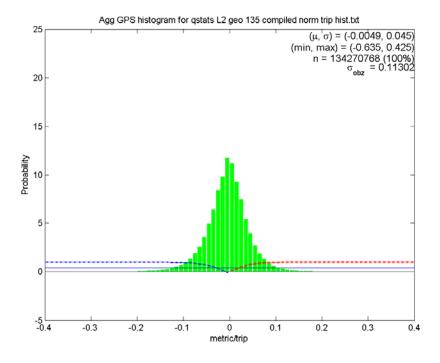


Figure 2.3-2 CCC GPS aggregate for L2

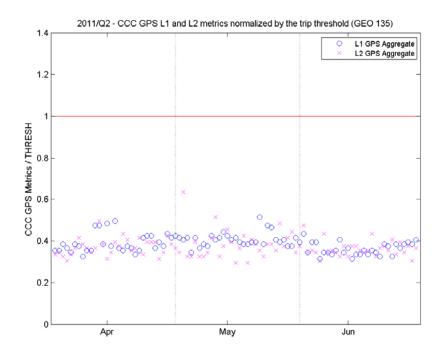


Figure 2.3-3 CCC GPS quarterly plot

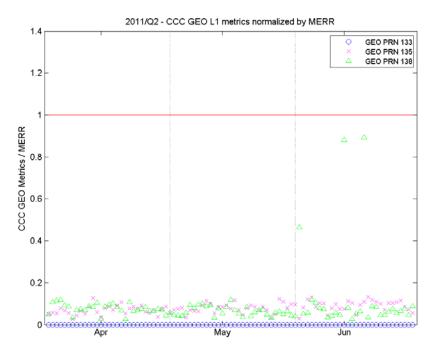


Figure 2.3-4 CCC GEO quarterly plot (integrity)

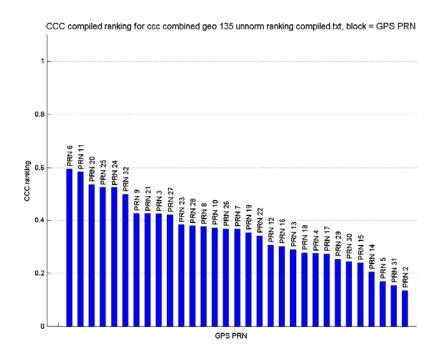


Figure 2.3-5 CCC GPS L1 L2 ranking plot sliced by PRN

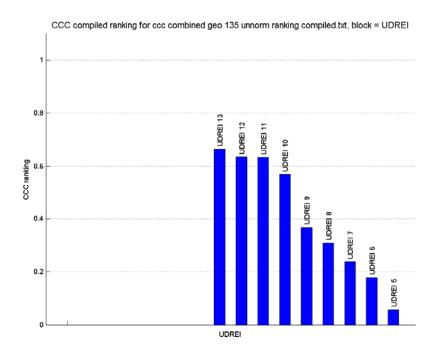


Figure 2.3-6 CCC GPS L1 L2 ranking plot sliced by UDRE

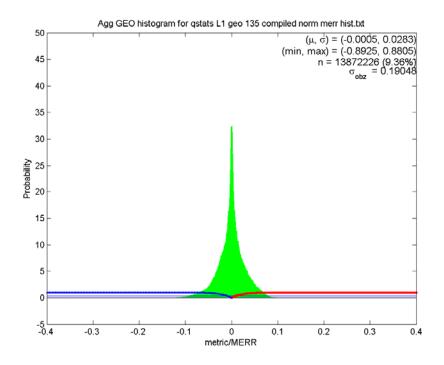


Figure 2.3-7 CCC GEO aggregrate plot for L1

2.4 Signal-quality monitoring (SQM)

SQM data is not available during this quarter due to processing problems. Results will be provided in the next quarterly report.

2.5 Satellite clock run-off analysis

No Satellite clock runoff events occurred during this quarter.

2.6 Ionospheric Threat Model

2.6.1 Daily percentage of GIVE Chi^2 values > 1

Note that data was unavailable between 5/24 through 6/6.

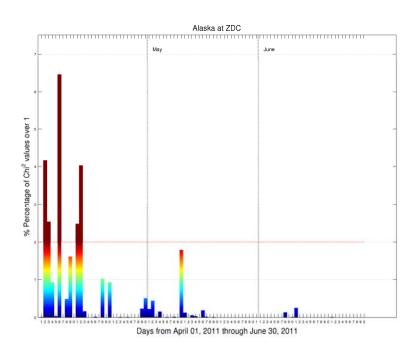


Figure 2.6-1 GIVE Chi^2 trips for Alaska on ZDC

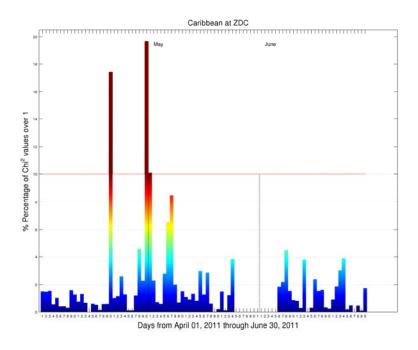


Figure 2.6-2 GIVE Chi^2 trips for the Caribbean on ZDC

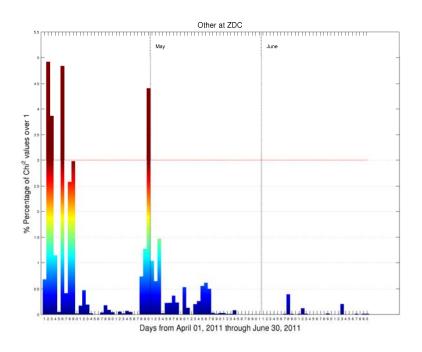


Figure 2.6-3 GIVE Chi^2 trips for Other region on ZDC

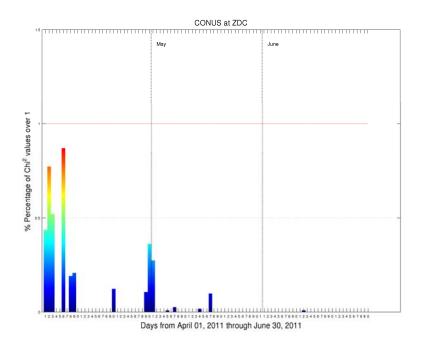


Figure 2.6-4 GIVE Chi^2 trips for CONUS on ZDC

2.6.2 Days of Interest

Table 2.6-1 Threat Model Days of Interest

Date of Trip	Alaska (2%)	Caribbean (10%)	Other (3%)	CONUS (1%)
2011-04-02	4.065		5.0202	
2011-04-03	2.5018		3.8743	
2011-04-06	6.5041		4.8001	
2011-04-11	2.5018			
2011-04-12	4.1097			
2011-04-20		17.4427		
2011-04-30		19.7285	4.417 9	

The days listed above require further analysis to determine if there are impacts to the Ionospheric Threat Model.

Days listed indicate regions where the percentage of Chi^2 values > 1 exceed the threshold value.

2.7 L1L2 Bias Levels

See supplemental material section.

3 Availability Monitoring

3.1 Service Volume Model

This analysis is under development.

3.2 System monitoring trips

Table 3.2-1 Reported Monitor trips

Monitor Trip	ZDC	ZLA	ZTL
BMV	0	0	0
ссс	12	6	12
L1L2	14	14	14
RDM (Threshold)	0	0	0
SQM	0	0	0
UPM	0	0	0
WNT	0	0	0
WRE BIAS	0	0	0

4 Continuity Monitoring

4.1 List of missed messages

The missed messages for the 2nd quarter of 2011 are displayed in the histograms in this section. Each histogram represents one GEO satellite. The histograms are annotated to briefly explain the cause for each instance of missed messages. The totals for missed messages per GEO satellite are as follows: CRW 135 - 67, CRE 138 - 27, AMR 133 - 553. Note that for the dates 5/27 - 6/6 data was unavailable due to power outages.

CRW missed messages for Q2 2011

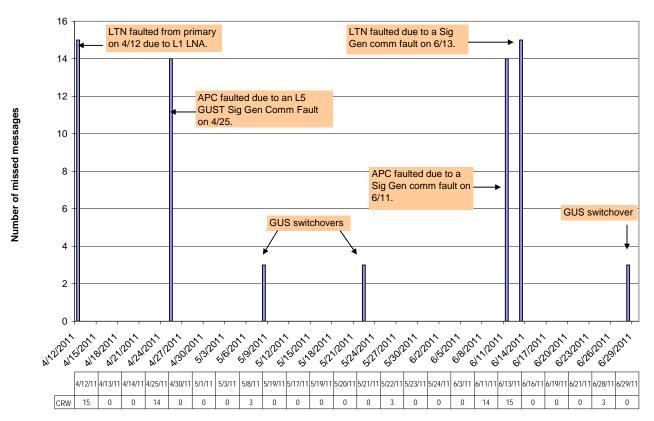


Figure 4.1-1 CRW missed messages Q2 2011

CRE missed messages for Q2 2011

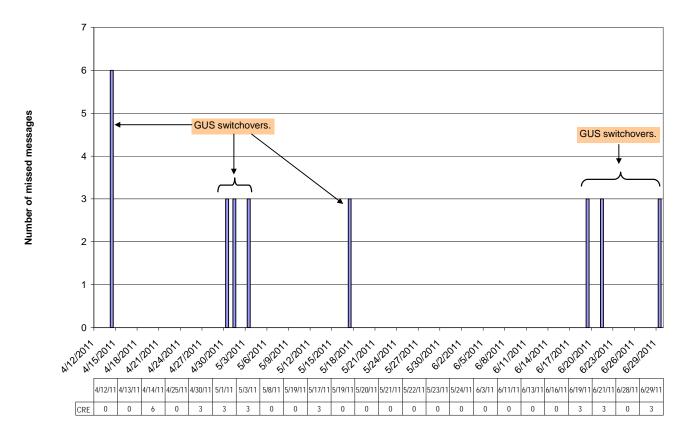


Figure 4.1-2 CRE missed messages Q2 2011

AMR missed messages for Q2 2011

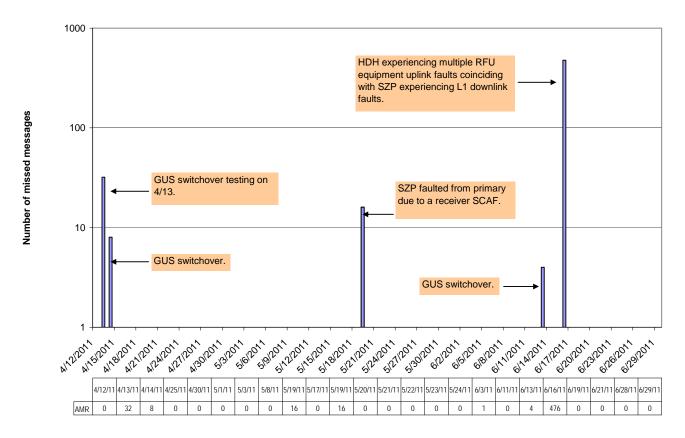


Figure 4.1-3 AMR missed messages Q2 2011

4.2 CNMP resets

This section will be added in a future revision.

5 Accuracy Monitoring

This section will be added in a future revision.

6 External Monitoring

6.1 Antenna phase center positions

- Results against CSRS-PPP indicate that rms position errors for all sites are less than 5 cm.
- Results against WFO-R2 indicate that rms position errors for all 107 sites are less than 5 cm.
 - o The 7 WREs with rms position errors greater than 5 cm are all less than 8.5 cm
 - o HNL exhibited the largest error, 8.5 cm

6.2 Ephemerides monitoring

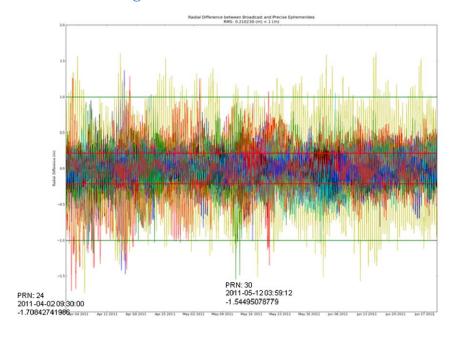


Figure 6.2-1Radial ephemeris

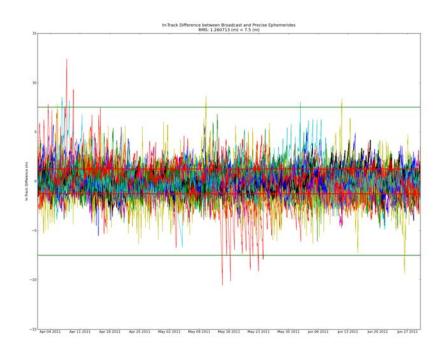


Figure 6.2-2 In-track ephemeris

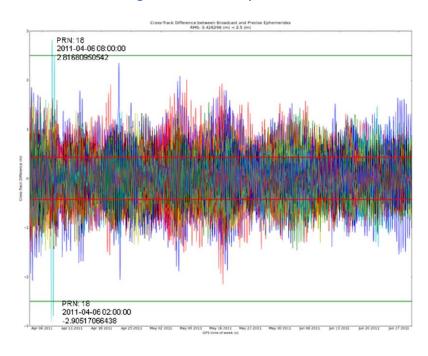


Figure 6.2-3 Cross-track ephemeris

Table 6.2-1 RIC outliers

PRN	R	I	С	PR	Ν	R	1	С
1	0	0	0	1	17	0	77	0
2	0	0	0	1	18	14	19	23
3	0	0	0	1	19	0	0	0
4	0	6	0	2	20	0	0	0
5	0	0	0	2	21	0	0	0
6	0	0	0	2	22	1	0	0
7	0	0	0	2	23	0	0	0
8	52	0	0	2	24	174	37	0
9	14	0	0	2	25	1	0	0
10	19	0	0	2	26	0	0	0
11	0	0	0	2	27	715	31	0
12	0	0	0	2	28	0	0	0
13	0	0	0	2	29	0	0	0
14	0	0	0	3	30	20	0	0
15	0	0	0	3	31	0	0	0
16	0	0	0	3	32	1	2	0

6.3 Space Weather Monitoring

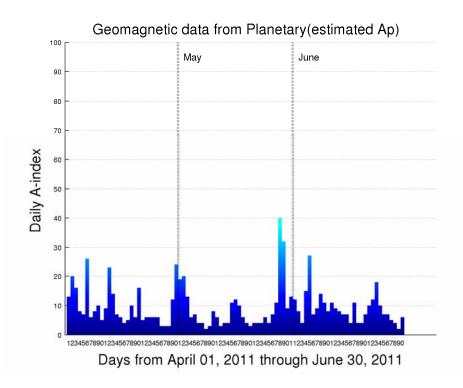


Figure 6.3-1 Daily estimated Planetary A-index

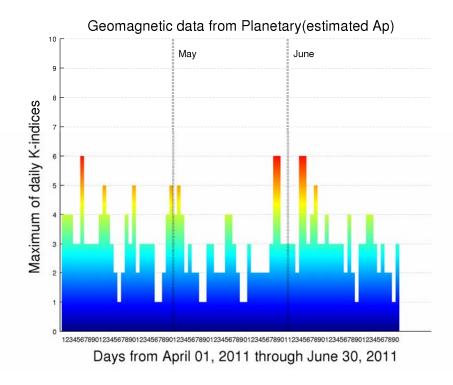


Figure 6.3-2 Maximum daily K-index from estimated Planetary

6.4 GEO CCC Signal Quality Analysis (SQA)

Two unique events caused degraded CCC on CRE during Q2. In early April, the Doppler change was excessively near 0 Hz each day. CCC performance improved after the Doppler magnitude increased over the same time period. Notably, PR oscillation correction would likely prevent this type of degradation; and hardware tests would be needed to verify software simulation.

In mid-June, the BRE GUST receiver experienced anomalous several-meter L5 PR jumps. The data and problem was analyzed by J. Lacey of Lockheed Martin and briefed at the weekly system monitoring teleconferences. The GUST receiver was replaced and since then the CCC statistic has exhibited nominal performance.

Figure 6.4-1 Doppler shift comparison for April 6th and nominal day

Figure 6.4-2 PRN 138 long-term CCC metric indicating effects of Doppler and BRE L5 PR spikes

7 Anomaly Investigation

7.1 AMR SIS outage in June

There was an AMR SIS outage that lasted approximately 7 minutes on June 16, 2011 beginning at 20:40:00 UTC. Paumalu (HDH) faulted first at 19:37:04 UTC and was in Maintenance mode at the time of a Santa Paula (SZP) GUST fault at 20:40:00. SIS was restored when Paumalu was put into Primary mode at 20:46:41 UTC. The GUST Significant Events related to the faults are listed in Table 7.1-1. Table 7.1-2 details the missing WUMs as there were displayed on the WAAS performance webpage.

Table 7.1-1 Sig Events detailing 2011-06-16 GUST Faults

06/16/2011	19:37:04	NEW	NO	NC	w	RN	GES	HDH	GUS-A	501	Mode	chan	ge to	Faulted
06/16/2011	19:37:20	NEW	NO	NO	WRN	GES	HDH	GUS-A	501	Mode	change	to	Maintenance	!
06/16/2011	20:40:00	NEW	NO	NO	WRN	GES	SZP	GUS-A	501	Mode	change	to	Faulted	
06/16/2011	20:41:51	NEW	NO	NO	INF	GES	HDH	GUS-A	501	Mode	change	to	Verification	ı
06/16/2011	20:46:34	NEW	NO	NO	INF	GES	HDH	GUS-A	501	Mode	change	to	Backup	
06/16/2011	20:46:41	NEW	NO	NO	INF	GES	HDH	GUS-	-A	501 Mo	ode cha	inge	to Pri	mary
06/16/2011 20:48:33 NEW NO NO WRN GES SZP GUS-A 501 Mode change to Maintenance														

Table 7.1-2 Missing WUMs during 2011-06-16 AMR SIS Outage

Missing	message	[06/16/2011	20:39:34	(992291989)	-	06/16/2011	20:39:35	(992291990)]	
Missing	message	[06/16/2011	20:39:36	(992291991)	-	06/16/2011	20:39:41	(992291996)]	
Missing	message	[06/16/2011	20:39:43	(992291998)	-	06/16/2011	20:39:46	(992292001)]	
Missing	message	[06/16/2011	20:39:47	(992292002)	-	06/16/2011	20:39:48	(992292003)]	
Missing	message	[06/16/2011	20:39:49	(992292004)	-	06/16/2011	20:39:53	(992292008)]	
Missing	message	[06/16/2011	20:39:56	(992292011)	-	06/16/2011	20:39:58	(992292013)]	
Missing message [06/16/2011 20:40:02 (992292017) - 06/16/2011 20:46:51 (992292426)]											

The SZP GUST fault at 20:40:00 UTC on this day was due to an SE 44 L1 GEO Downlink Message Check Fault. This significant event was caused by low C/No which in turn caused the GUST receiver at SZP to lose lock on the L1 GEO downlink signal. After 60 seconds of missed WUMs on its downlink signal, SZP faulted from Primary mode. The root cause of the low C/No was commercial power problems at SZP which caused the antenna to stop tracking.

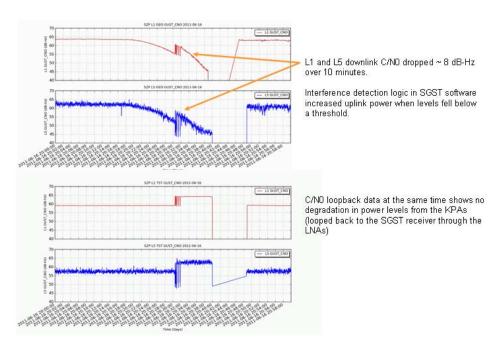


Figure 7.1-1 SZP Downlink C/No Degradation prior to fault on 2011-06-16

Paumalu faulted 8 times during the day prior to the fault at 19:37:04 and was kept in Maintenance to accommodate the replacement of a faulty Radio Frequency Uplink (RFU) Frequency Distribution Amplifier (FDA) by site technicians. In each case the HDH Maintenance and Control Computer (M&C) reported FDA faults and replacement of the RFU FDA did not resolve the issue.

Later investigation of the HDH FDA showed that the faults were characterized by the M&C loss of Ethernet communication with the FDA. When this happened, a fault was raised and a 30 second recover timer began. After 30 seconds, the M&C attempted to re-establish the communication socket. Each communication failure was brief and had been observed to last less than 12 seconds. After each failure, communication was reestablished in less than 12 seconds. The FDA communication failures did not adversely affect the fundamental operation of the FDA frequency distribution as it continued to pass the proper frequency references to the upconverters and test loop translators in the RFU. The failures caused the RFU to pass a communication fault alert to the HDH Signal Generator Subsystem (SGS) which then declared a GUST Fault. This type of FDA fault was experienced previously in 2009 on the Paumalu Broadband Global Area Network (BGAN) RF Subsystem. At that time, the site worked with the FDA manufacturer but was unable to develop a fix and instead a change was made to the M&C application's alarming criteria. An M&C workaround was implemented to designate the FDA communication faults as a warning rather than a major alarm. The alarm was made visible to maintainers but was no longer reported to the SGS. This M&C update was deployed to both AMR GUSTs as an emergency mod in September 2011. HDH was upgraded on 2011-09-01 and SZP was upgraded on 2011-09-13 and the issue is now being closely monitored.

8 Materials and Methods

This section describes which tool was run, flags set, build number, data files used, any pre-processing done, etc.

8.1 Antenna Positioning

- Sites were surveyed on 2011-04-20 and cross compared against the following:
 - o CSRS-PPP
 - o WFO-R2
 - o Coordinates projected to six months beyond WFO WFO Release 3, that is, 2012-05-01
 - used latest page-NT version (k)
 - NGS updated to IGS05 coordinates
- Accurate antenna positions needed to support DGPS applications
- Correct for Time Dependent Process
 - o Tectonic Plate Movement
 - o Subsidence
- Correct for Shift Events
 - o Seismic
 - o Maintenance
- WIPP Review for integrity issues
 - o Greater than 10 cm WIPP should review
 - o Greater than 25 cm WIPP must review
 - Special case for Mexico City (25 cm for review)
- Project the need for a WAAS Antenna Coordinate Update

8.2 Clock Runoff

8.2.1 Monitoring approach

- Events typically result in a fast correction that exceeds 256 meters
- When this occurs, the satellite is set Do Not Use until the correction reaches a reasonable size
- Events where the satellite is set Do Not Use from excessively large fast corrections while the satellite is healthy are recorded

8.3 Ionospheric Threat Model

8.3.1 Monitoring approach

- Monitor for Chi² values greater than 1 in the four regions, including:
 - o CONUS > 1%
 - o Alaska > 2%
 - o Caribbean > 10%
 - o Other > 3%

8.4 Ephemeris

Compared broadcast vs precise in HCL to ensure sigmas are less than 1m, 2.5m, 7.5m for Radial, Cross Track, and In Track.

8.5 GEO Signal Quality Analysis

8.5.1 Data

The analysis uses MATLAB to compile data from the GEO monitoring equipment in OKC.

8.5.2 Tools

The MATLAB analysis and plotting code is run on machine provided by Zeta.

9 Supplemental Material

9.1 Ionospheric Threat Model

9.1.1 Monitoring regions

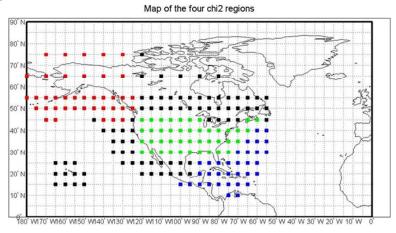


Figure 9.1-1 Map of the four Chi^2 regions

9.2 CNMP

The assertion in the HMI documents for CNMP states:

The Code Noise and Multipath (CNMP) error bound is sufficiently conservative such that the error in linear combinations of L1 and L2 measurements is overbounded by a Gaussian distribution with a sigma described by the Root Sum Square (RSS) of L1 and L2 CNMP error bounds except for biases, which are handled separately.

To monitor this assertion, we check the bounding for three statistics, L1, IFPR, and Delay. The equations used to determine a passing or failing grade for the distribution plots are in the equations appendix. The sigma zero-centered sigma overbound plots are considered to be passing if the value is less than one, which is marked in the plots. For the second quarter of 2011, there were no failures for the sigma overbound or any of the aggregate or sliced distribution plots.

9.2.1 Analysis on poor performing sites

For the second quarter of 2011, there was one high point on the daily aggregate sigma overbound plot (Fig 2.1-4). This point was high due to MPR-A, RDL2. The distribution plot sliced for this WRE can be seen below.

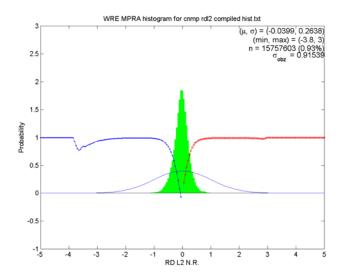


Figure 9.2.1-1 MPR-A Poor CNMP Performance

There is no known reason for the poor performance at this site. The performance of this site for the third quarter of 2011 will analyzed to determine whether this was just an isolated event or is part of a long term trend.

MSD-A has had issues in the past that were believed to be due to bad cabling at the site. The analysis from this quarter confirms that the cabling was the issue. The performance at this site with the new cable was nominal for the entire quarter. This is illustrated in Figure 9.2.1-2.

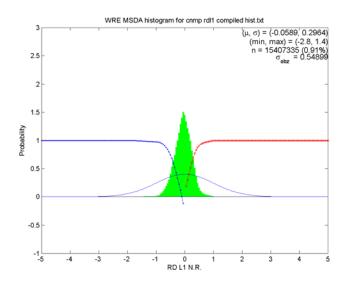


Figure 9.2.1-2 MSD-A Nominal CNMP Performance

The following Table contains more detailed information on the worst performing sites of the quarter. This Table is generated by ranking the worst threads for each of the three statistics. The worst thread in a category was given 5 points, while the second worst was given 4 points, and so on. The points were totaled across all three metrics to give the table below. Only one thread per WRS appears below, so that 5 WRS are on the table (i.e. all three Houston threads might otherwise appear in the table).

Sliced by WRE Legend: - = passed	L1				IFPR				Delay			
X = did not pass WRE#, WRE Name	μ	σ	max	pass/ fail	μ	σ	max	pass/ fail	μ	σ	max	pass/ fail
106, Tapachula_B	-0.058	0.35	4.05	•	-0.032	0.31	2.65	•	0.017	0.32	3.20	-
29, Houston_C	-0.072	0.44	2.65	-	0.105	0.41	2.80	•	-0.110	0.40	2.80	-
6, Anchorage_A	-0.050	0.28	1.90	-	-0.012	0.28	2.60	-	-0.006	0.29	2.65	-
64, Seattle_B	0.004	0.26	3.45	-	-0.004	0.24	2.50	-	0.073	0.24	1.85	-
57, Oakland_A	-0.021	0.22	1.45	-	0.010	0.21	1.50	1	-0.023	0.21	1.60	-

Table 9.2-1 CNMP results from poor performing WRE slices

9.3 GEO Signal Quality Analysis

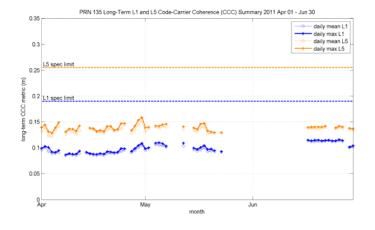


Figure 9.3-1 long-term CCC PRN135

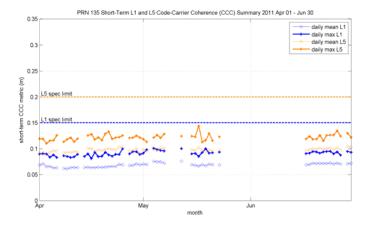


Figure 9.3-2 short term CCC PRN 135

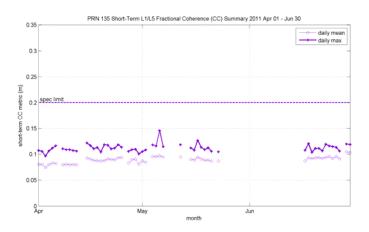


Figure 9.3-3 PRN 135 short-term CC

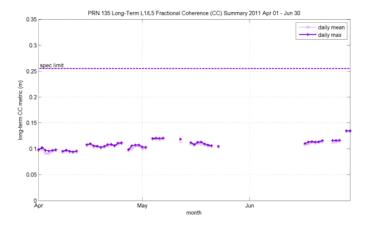


Figure 9.3-4 PRN 135 long-term CC

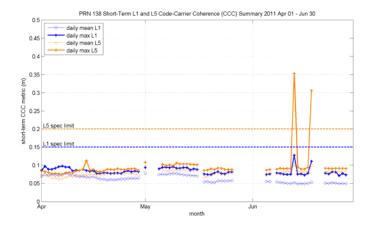
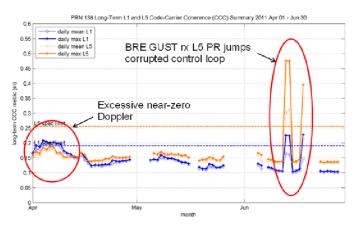


Figure 9.3-5 PRN 138 short-term CCC



Note: missing values indicate days with switchovers or incomplete data

Figure 9.3-6 PRN 138 long-term CCC

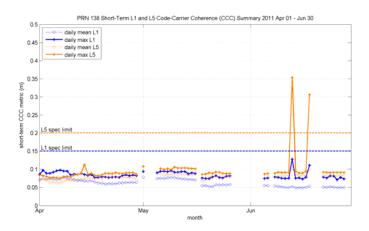


Figure 9.3-7 PRN 138 short-term CCC

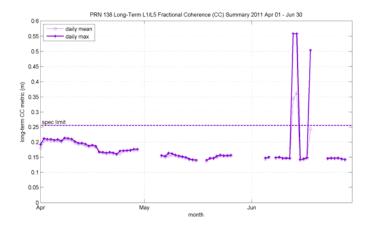


Figure 9.3-8 PRN 138 long-term CC

9.4 L1L2 Bias Levels

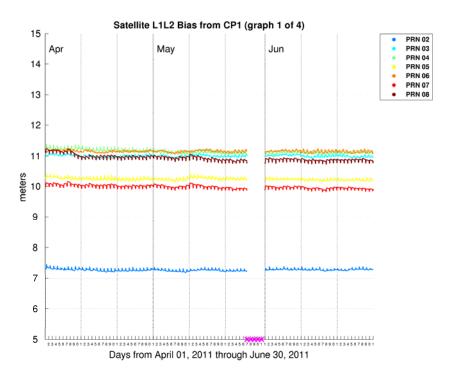


Figure 9.4-1 PRN2-8 L1L2 bias from CP1

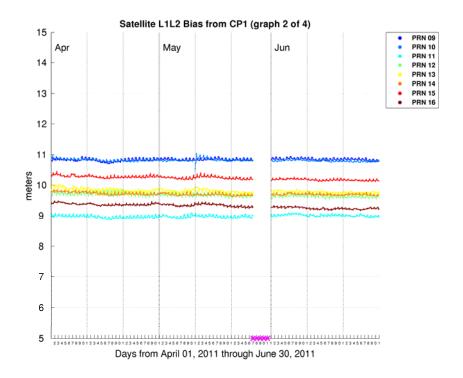


Figure 9.4-2 PRN9-16 L1L2 Bias from CP1

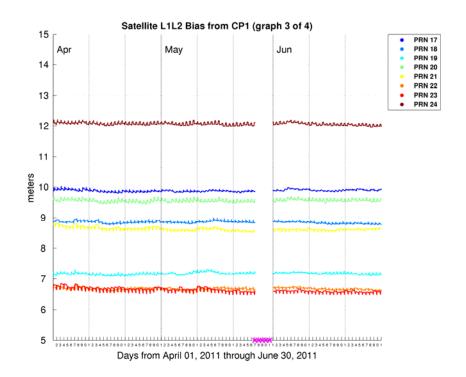


Figure 9.4-3 PRN17-24 L1L2 Bias from CP1

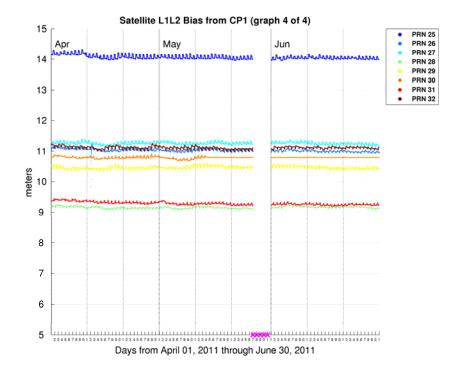


Figure 9.4-4 PRN25-32 L1L2 Bias from CP1

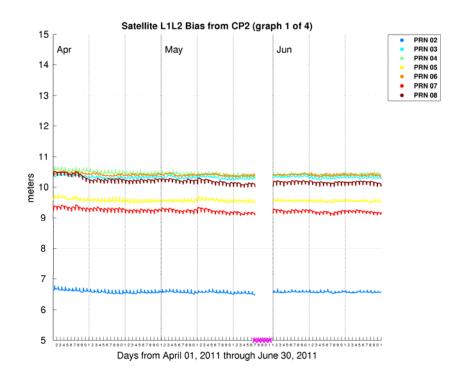


Figure 9.4-5 PRN2-8 L1L2 Bias from CP2

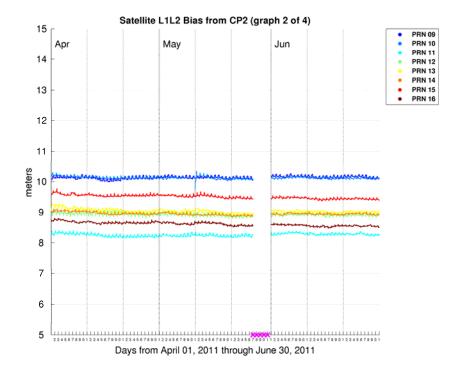


Figure 9.4-6 PRN9-16 L1L2 Bias from CP2

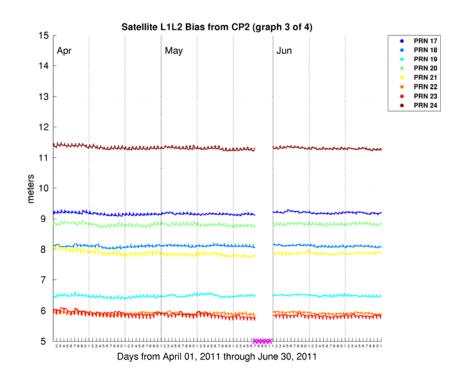


Figure 9.4-7 PRN17-24 L1L2 Bias from CP2

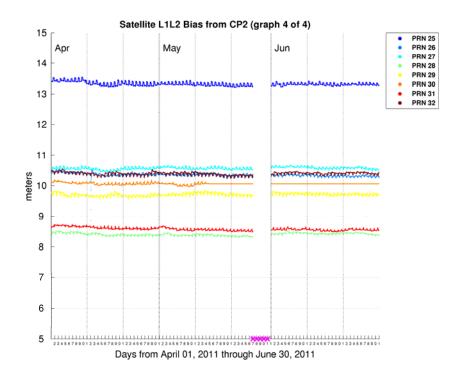


Figure 9.4-8 PRN25-32 L1L2 Bias for CP2

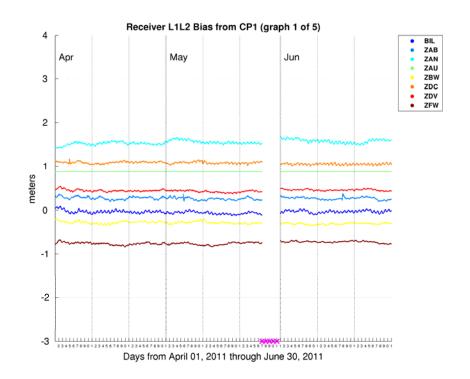


Figure 9.4-9 Receiver L1L2 Bias from CP1 (1 of 5)

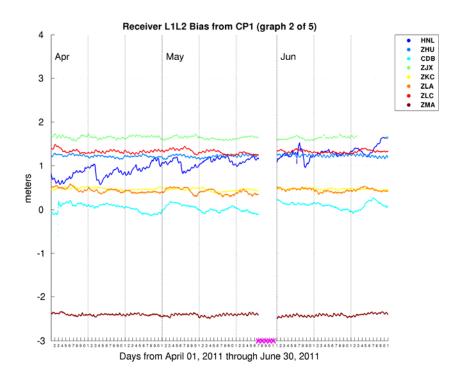


Figure 9.4-10 Receiver L1L2 Bias from CP1 (2 of 5)

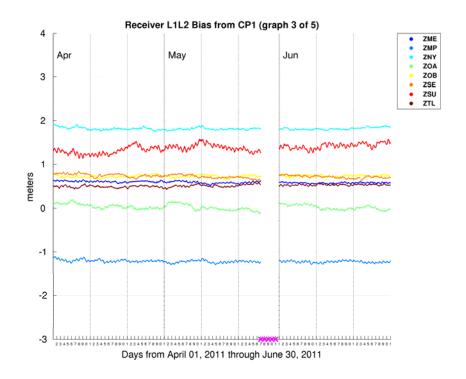


Figure 9.4-11 Receiver L1L2 Bias from CP1 (3 of 5)

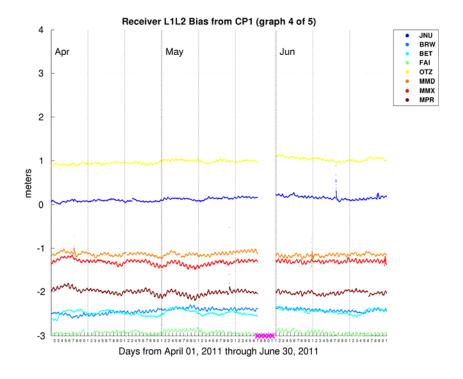


Figure 9.4-12 Receiver L1L2 Bias from CP1 (4 of 5)

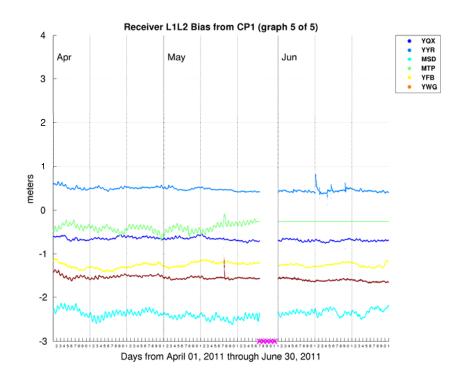


Figure 9.4-13 Receiver L1L2 Bias from CP1 (5 of 5)

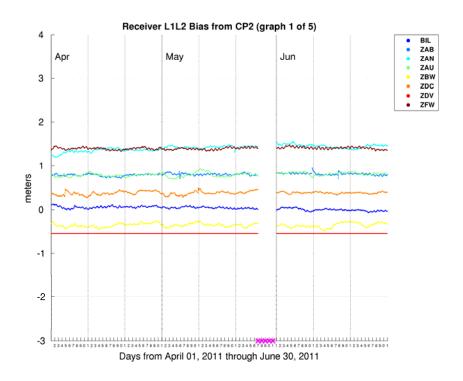


Figure 9.4-14 Receiver L1L2 bias from CP2 (1 of 5)

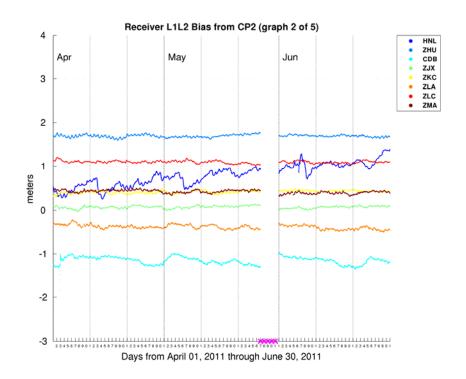


Figure 9.4-15 Receiver L1L2 bias from CP2 (2 of 5)

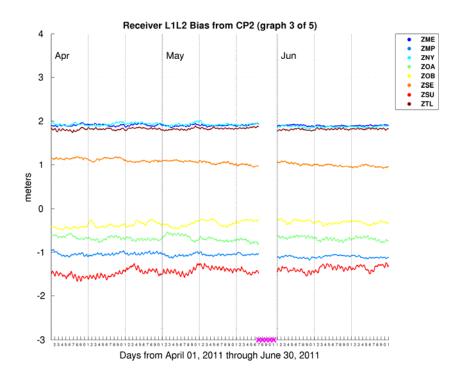


Figure 9.4-16 Receiver L1L2 bias from CP2 (3 of 5)

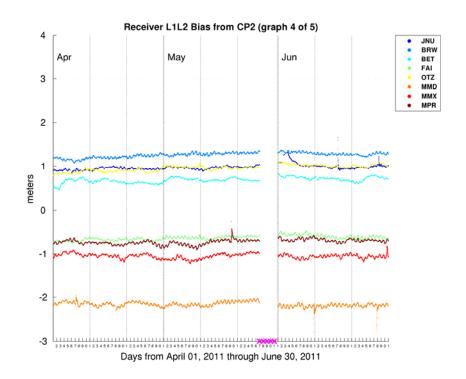


Figure 9.4-17 Receiver L1L2 bias from CP2 (4 of 5)

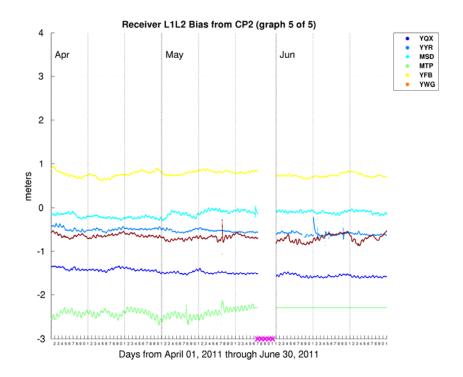


Figure 9.4-18 Receiver L1L2 bias from CP2 (5 of 5)

10 Appendix

10.1 Equations

10.1.1 CNMP

The cumulative density function (CDF) is defined as:

$$\Phi^R(x) = \frac{1}{\sqrt{2\pi}} \int_x^{\infty} e^{-t^2/2} dt$$

Equation 10.1-1: Cumulative Density Function (CDF)

$$\Delta(x) = \begin{cases} \frac{\Phi_{theory}^{R}(x) - \Phi_{data}^{R}(x)}{\Phi_{theory}^{R}(x)} & x \ge 0\\ \frac{\left[1 - \Phi_{theory}^{R}(x)\right] - \left[1 - \Phi_{data}^{R}(x)\right]}{\left[1 - \Phi_{theory}^{R}(x)\right]} & x < 0 \end{cases}$$

Equation 10.1-2

CNMP passes for $\Delta x > 0$ for all |x| > 0.25.

10.2 References

WAAS CDRL A014-011 "Algorithm Contribution to HMI for the WAAS"

10.3 Assertions

10.3.1 Code Carrier Coherence

The *a priori* probability of a CCC failure is less than 1x10-4 per set of satellites in view per hour for GPS satellites and 1.14x10-4 for GEO satellites.

10.3.2 CNMP

The Code Noise and Multipath (CNMP) error bound is sufficiently conservative such that the error in linear combinations of L1 and L2 measurements is overbounded by a Gaussian distribution with a sigma described by the Root Sum Square (RSS) of L1 and L2 CNMP error bounds except for biases, which are handled separately.

10.3.3 Antenna Monitoring

The position error (RSS) for each WAAS reference station antenna is 10 cm or less when measured relative to the ITRF datum for any given epoch. (Mexico City is allowed 25cm). The ITRF datum version (realization) is the one consistent with WGS-84 and also used for positions of the GPS Operational Control Segment monitoring stations.

10.3.4 Ephemeris

The CDF of GPS ephemeris errors in a Height, Cross-track, and Along-track (HCL) coordinate system is bounded by the CDF of a zero-mean Gaussian distribution along each axis whose standard deviations are σosp-ephh, osp-

ephc, and ②osp-ephl. The probability that a satellite's position error is not characterized by this a priori ephemeris model is less than 10-4 per hour.

10.3.5 Ionospheric Threat Model

The values of $\sigma_{\text{decorr_undersampeled}}$ and $\varepsilon_{\text{iono}}$ adequately protect against worst case undersampled ionosphere over the life of any ionospheric correction message, when the storm detectors have not tripped.

10.4 Coding standards and guidelines

The standards and guidelines for the Offline Monitoring effort are recorded here. "Standards" represent a "rule" that is assumed to be "enforceable", that is, it has been agreed to by the stakeholders and recorded as official. PCRs can (but not necessarily will) be blocked due to lack of upholding a standard. Furthermore, standards can certainly have exceptions, but these are dealt with on a case-by-case basis and recorded as such. "Guidelines", on the other-hand, are not enforceable. Guidelines represent good ideas and common engineering practices across the group. PCRs cannot be blocked as a result of not following a guideline.

Transitioning from a guideline to a standard is a done on a case-by-case basis. While there is no hard and fast rule for how this is done, the steps for this usually contain an initial agreement by the stakeholders (which included management and engineers) that a standard ought to be adopted, a resource (with associated level of effort) assigned, and an initial assessment as to how much work is involved (estimated end date, etc). The process of transitioning from a guideline to a standard is known as refactoring, and the practice is encouraged as long as stakeholder buy in is considered at each step.

The standards and guidelines are differentiated by the words "shall" and "should".

10.4.1 Integrity standards for MATLAB

The integrity standards for MatLab were developed during the WAAS FLP Release 6/7 time frame. These standards represent rules that, if broken, could lead to incorrect or erroneous results (not necessarily a tool crash but actual incorrect output). These are documented in the WAAS HMI document (in section 4.3.3 of that document) and are repeated here in their condensed form. More detail can be found in the WAAS HMI document. Note that these standards are enforced by use of the CD_STD_CHK tool which parses the files/scripts line by line checking for breaches.

- MATLAB Calling Ambiguity:
 - o Ensure that no MATLAB keywords are used as function names.
 - Use functions, not scripts.
 - o Function name and filename being the same is required.
 - One function per file required.
- Functions should not be influenced by anything other than inputs:
 - o No global variables.
 - No persistent variables.
- MATLAB Functionality Ambiguity:
 - o The **squeeze** function shall not be used.
- Windows Ambiguity:
 - The exist function shall not be used.

- Coding Clarity:
 - o The **eval** command shall not be used.
- Consistency Check:
 - OSP consistency must be addressed.
 - Critical parameters need to not be hardcoded in the tools
- Repeatability:
 - The actual scripts that were used to generate the data, tables and plots need to be captured along with the outputs, as well as a mapping to the actual data set used.

10.4.2 Offline Monitoring Coding Standards

Along with the Integrity standards described in section 9.4.1, there exist several "Offline Monitoring" coding standards. These are coding standards which are attached to the running of the Offline Monitoring code and which have been identified as required for the processing of the offline monitoring data. Currently, there are five standards

- All open files shall be closed
 - This requirement should be applied over all tools for all Offline Monitoring scripts. This
 requirement is simple, as it just requires that any file which is opened be appropriately closed in
 the same script that opens it.
- In MatLab, the "figure" command needs to always have a file ID associated with the open figure
 - The MatLab coding language allows the user to create "figures" without assigning a file id variable. Closing the specific figure is then impossible in general, and the figure must be closed either by keeping track of the "current" figure ID, or by applying the "close all" command. Neither of these is desired, and as such, each figure must have a unique file ID in memory.
- In MatLab, the "close all" command shall not be used.
 - The close all command is issued to close all figures with or without a file ID. As standards are in place to assign a file ID for all figures, this line of code is unnecessary and should not be used.
- All open figures should have the option to be closed
 - The MatLab tools should not leave open figures after the analysis is run (by default). For particular tools, it may be desirable to keep the plots up on the screen, but the option to close them should be implemented
- Use cs saveplot for saving plots in MatLab
 - The cs_saveplot function is a common script which saves figures to results directories. There are several options when saving a plot, as using this function allows one place to modify the saving requirements.

10.4.3 File naming conventions

While no complete convention exists, there are standard "pieces" which shall be enforced for the Offline Monitoring effort. These refer to the labels inside the actual name of the tool which refer to information in the data file. The requirements are listed below.

• Filenames shall be named using a prefix, followed by an "_", then the ISO8601 date in the form of YYYY-MM-DD, followed by a "." and the extension.

- Filenames shall use lowercase letters, integers, underscores and dashes.
- There shall be no more than one "." in a file name
- Text files shall end with the suffix ".txt"
- Binary files shall end with the suffix ".bin"
- Files which contain data for a particular PRN shall have a six-character label of the form "prnDDD" where DDD are the three digits referring to the PRN number. PRNs less than 100 shall have a leading zero, and PRNs less than 10 shall have two leading zeros.
- Files which contain data for a particular WRE shall have a six-character label of the form "wreDDD" where DDD are the three digits referring to the WRE number. WREs less than 100 shall have a leading zero, and WREs less than 10 shall have two leading zeros. Also note that WREs start counting at 0, so for a 38-station system, the WRE number range from 0 to 113.
- Files which contain data for a particular UDREI shall have a seven-character label of the form "udreiDD" where DD are the two digits referring to the UDRE index. UDREIs less than 10 shall have a leading zero. Also note that UDREIs start counting at 0, so UDREIs range from 0 to 15.
- Files which contain data for a particular GIVEI shall have a seven-character label of the form "giveiDD" where DD are the two digits referring to the GIVE index. GIVEIs less than 10 shall have a leading zero. Also note that GIVEIs start counting at 0, so GIVEIs range from 0 to 15.

10.4.4 Histogram slicing and bin requirements

For many of the analyses, histograms are used to show compliance to particular requirements. As there is inherent averaging in the creating an aggregate histogram, the concept of slicing was introduced early in the WAAS analysis process. This requires that data from (potentially) different error sources are not averaged into a single histogram, but are examined separately. In order to compile results across multiple days (and data sets), both the bin centers and the number of columns for each type of slice needs to be fixed. Modifying these requirements at a later date would make long term trending difficult, if not impossible.

Table 10.4-1 below shows the bin requirements for the data files which are to be histogrammed by one or more of the Offline Monitoring analyses. Note that the minimum and maximum data cutoffs are defined to be the bin EDGES, not the bin centers. Thus, the bin centers are in between the edges defined in table 10.4-1.

Data file Bin width **Data Description** Data min Data max Units Raw CCC metric (L1 and L2) Qstats* -8.0 0.01 8.0 meters CCC metrics normalized by -3.0 0.01 3.0 Qstats* None trip threshold CCC metrics normalized by -2.0 2.0 Qstats* 0.001 None MERR 0.001 Max SQM metric SQM_reduced 0 2.0 None

Table 10.4-1 Data histogram bin requirements

Table 10.4-2 below shows the slicing requirements, i.e., the number of columns (and their designations) for each type of slice.

Table 10.4-2 Slicing requirements

Slice Description	# columns	Column Description
Aggregate	1	This is the histogram of the entire metric. There is always one column, no more.
UDRE index	16	Columns 1 through 14 represent the data associated with a UDREI of one less than the column, i.e., UDREIs of 0 through 13. The last two columns represent satellites which are NM and DU respectively.
PRN	44	The PRN slices come in a few sets. The first set is the first 32 PRNs. The second set is 10 columns devoted to past, current and future GEOs. The first five GEO columns are the GEO PRNS of 122, 133, 134, 135, and 138. The next five columns are reserved for future GEO PRNS. Finally, the last two columns are the aggregate of the GPS data ns the aggregate of the GEO data respectively. This order is maintained.

10.4.5 OLM file formats

Standard file formats have been defined for four types of files, listed below. These represent standards, and are enforceable requirements.

10.4.5.1 Histogram files

The number of columns in a histogram file shall be one more than the sum of the number of slices. For example, is a histogram file contained an aggregate histogram, slices by UDREI and slices by PRN (both GEO and GPS), there would be 1+1+16+44 = 62 columns. The first column is the bins, the second column is the aggregate, columns 3 through 18 are the 16 UDRE slices (with columns 17 and 18 being NM and DU), columns 19 through 50 are the 32 GPS PRNs, columns 51 through 60 are the GEO PRNS (which the last five being held in reserve), column 61 is the aggregate GPS histogram and column 62 is the aggregate GEO histogram.

- Histogram files are stored as raw counts, not probabilities and the bins are recorded as bin centers.
- Histogram files can be daily or compiled into a report.
- The histogram file shall have a header which has column headings lined up with the columns of the data.

10.4.5.2Statistics files

Each statistic in the statistics file shall be defined to be able to be computed using bins (either centers or edges) and the raw counts, and each column in the histogram file shall have all statistics computed for it. Thus, the dimensions of a statistics file shall be as such.

- The number of rows is the same as the number of statistics
- The number of columns shall be the same as the number of slices

In order to account for the column of bins, a statistic index is placed there, so that each column in a histogram file corresponds to the same column in the statistic file. There are currently fifteen descriptive statistics computed for each histogram file:

- 1. Counts
- 2. Mean
- 3. Standard Deviation
- 4. Minimum
- 5. Maximum
- 6. Absolute Maximum
- 7. Sigma Over-bound (Zero-centered)
- 8. Sigma Over-bound (Mean-centered)
- 9. 1st Quartile
- 10. Median (2nd Quartile)
- 11. 3rd Quartile
- 12. Mean of Absolute Value
- 13. Standard Deviation of Absolute Value
- 14. RMS
- 15. Variance

The statistics file shall have a header which has column headings lined up with the columns of the data, as well as the list of statistics represented in the file.

Statistics files can be daily or compiled into a report.

10.4.5.3 Time Series files

Time series files represent a quantity which evolves over time. These can be any quantity, but currently only satellite quantities are created. Thus, the file naming convention for PRN (described in 4.4.2) are utilized.

The time series files have as the first three columns three different representation of time. The first is WAAS time, the second is UTC in ISO-8601 format (HHMMSS) and the third is seconds in the day.

After the first three columns, more columns can be added. The intent of the time series file is to have all of the data which a plot would require in the subsequent columns.

Time series files are only attached to daily quantities, but several time series files could be concatenated together to create a multi-day files (and plot).

10.4.5.4Quantity files

Quantity files contain two dimensional slices of a particular quantity. For example, creating a UDREI/ GPS PRN slice for the absolute maximum of the CCC metric would allow a user to see which satellite have issues at which UDREIs. As both dimensions are used, only one statistic per file can be represented.

Quantity files are currently only daily files, but they could be created for a compiled data for some statistics.

10.4.5.5Quarterly files

Quarterly files are the files which are plotted over the period of the quarter. Thus, the first column is the number of the day in the quarter and the second (and subsequent) columns are data to be plotted. The data set can be customized for the particular plot.

10.5 Offline Monitoring Process and Procedures

10.5.1 Schedule and Meetings

The offline monitoring group will meet approximately twice a quarter. One set of meetings is to be set for the first week of the new quarter to go over plans for that quarter. The second set of meetings is to be set for shortly before the WIPP. For both meetings, the general purpose is to plan for the next WIPP or the next OLM report, as the case may be. At the meetings, task lists with priorities and resources are created, to be reviewed at the next set of meetings.

The offline monitoring document is release once a quarter. The analyses should be running during the quarter, and should be being reviewed on a periodic basis. Once the quarter ends, three dates are pertinent.

- Two weeks after the quarter ends All analyses complete
- Four weeks after the quarter ends Draft document released
- Six weeks after the quarter ends Final document completed

10.5.2 Data Processing

The data processing strategy for the Offline Monitoring document is to currently run the safety processor prototype on blocks of snoop files, approximately one week long. The blocks are then run in succession to create a "quarters" worth of data, which spans the three months of the quarter in question. The blocks of data are usually a week long, but due to data issues, as well as week VS month cutoff issues, the lengths of the individual blocks may vary.

Standard processing is applied across the analyses for the individual days. This includes the creation of histogram files, histogram statistics files, time series files, and two dimensional quantity files. There are associated plots as well for each of the above mentioned plots. In addition to the standard processing, analyses specific to the tool are also run for each day. In this way, analysis specific data reduction and results are generated on a daily basis.

Once the daily analyses have been run, the results are compiled into a "report" directory. This includes the accumulation of histogram data, and the plotting of statistics across the quarter.

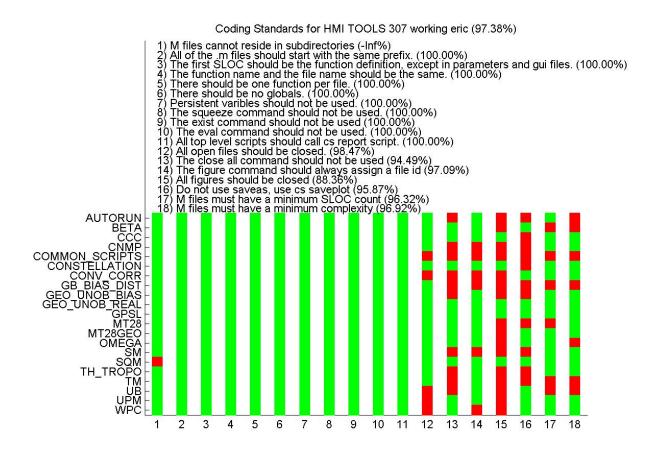
10.5.3 Tool Strategy

Tool builds created at both SOS and SRC are valid, and need to have proper versioning attached to them. All of the results from a single quarter should come from one version of a tool, and this version should be recorded in the OLM document.

Both regression testing and coding standards checking are as automated as possible, and both have tools associated with them. For the regression testing, the "reg" MatLab tool has been created. This tool is stored in the OLM repository, and runs the regression tests for the MatLab tools in an automated way (from reg_go). The coding standards are checked via the CODE_STD_CHK tool. There is one standard which checks that all of the scripts are in the top-level directory, followed by the ten integrity standards, followed again by the five OLM coding standards.

As is often the case, tools (old and new) do not comply with the coding standard at the outset. As such, a "refactoring" approach is adopted. By "refactoring", it is mean that some way to assess the level of non-compliance is required (either by manual review or via automation) before work commences on fixing the issue across the tool set. Once this is assessed, the work commences as is best seen fit by the group, and the standard is enforced for future tools.

An example of the current coding standards plot is shown below. As can be seen, the SQM tool is the only tool which does not have all of its scripts in the top level folder. Thus, it is not possible to assess any other issues until that first issue has been worked. For the other tools, the ten integrity standards are all met, and then several of the OLM standards are in a state of non-compliance. As of the writing of this document, PCRs are in place to fix the issues. Note that two other standards (which have to do with SLOC count and complexity) are also listed. The plot below just serves as an example.



10.6 Acronym and Abbreviations

==Numbers==
1PPS One Pulse per Second
==A==
AAC Aeronautical Administrative Communication
A-S Anti-spoofing
A/D Analog-to-digital (conversion)
AC Alternating Current
ACEPS ARTCC Critical and Essential Power System
ACS Access Router
ADC Analog-to-digital converter
ADD Algorithm Description Document
ADR Accumulated delta range
ADS Automatic dependent surveillance
ADS_B Aircraft—automatic Dependent Surveillance-Broadcast
AF Airway Facilities (Airway Facilities name has been changed to Technical Operation)
AGC Automatic gain control
AGL Above Ground Level
AHRS Attitude and heading reference system
AlC Akaike information-theoretic criterion
AIRS Advanced inertial reference sphere
AIS Aeronautical Information System
AIT Automatic Integrity Test
AlX Advance Interactive eXecutive
ALF Atmospheric loss factor
ALS Autonomous landing system
altBOC Alternate binary offset carrier
ANH Access Node Hub
AODE Age of data word, ephemeris
AOR-E Atlantic Ocean Region East (WAAS)
AOR-W Atlantic Ocean Region West (WAAS)
AOS Operational Support Navigation and Landing Branch (PNB.1.2.2.2 p.35)
APV-I Approach with Vertical Guidance (For APV-I, VAL = 50m, HAL = 40m in HMI)
AR Autoregressive
ARMA Autoregressive moving average
ARNS Aeronautical Radio Navigation Services
ARP Antenna Reference Point
ARTCC Air Route Traffic Control Center
ASD Amplitude spectral density
ASIC Application-specific integrated circuit
ASL Above Sea Level
ASQF Application-Specific Qualification Facility (EGNOS)
ATC Air Traffic Control
ATCT Air Traffic Control Towers
ATCU Air Traffic Control Unit
ATM Air Traffic Management
AWACS Airborne Warning and Control System
==B==
BCN Backbone Concentrator Node
BDUMP Binary Data Utilities and Manipulation Program
BEI Bias Error Injection (HMI)
BIPM International Bureau of Weight and Measures
BIT Built-in Test
BMV Broadcast Message Validation

BOC Binary offset carrier

BOE Basis of Estimate
Bps Bits per Second

BPSK Binary phase-shift keying

BQT Build Qualification Test

BRASS Bilingual Request and Security Subsystem

BUTS

BW Bandwidth

==C==

C&V Correction and verification (WAAS) C/A Coarse acquisition (channel or code)

C/No Carrier to Noise

CASE Computer Aided Software Engineering (it is a Software Engineering System tool)

CAT Collection & Analysis Tools

CCB Configuration Control Board (hardware only - see SCCB for Software [PNB.1.2.2.4, p4]

CCC Code Carrier Coherence

CCD Configuration Control Decision

CCS Configuration Control System

CDCV

CDF Cumulative Distribution Function
CDM Code-division multiplexing
CDMA Code-division multiple access
CDTI Cockpit Display of Traffic Information

CDRL Contract Data Requirements List

CEP Circle error probable

CHIA Change Impact Assessment aka Change Impact Analysis

CID Commercial Item Drawing

CIDS Critical Item Development Specification

CLIN Contract Line Item Number

cm centimeters

CMCI Code Minus Carrier Minus Iono Delay

CME coronal mass ejection (when the sun emitted a C-class solar flare in space that effect on WAAS)

CMP Comparator (Fault Tree)
CMTP Contractors Master Test Plan

CNMP Code Noise and Multipath (CNMP) | Code Noise and Multipath

CNS Communication Navigation Surveillance

COB Close of Business

CONUS Conterminous United States

Contiguous United States

Continental United States

CORS Continuously Operating Reference Station

COSPAS Acronym from transliterated Russian title "Cosmicheskaya Sistyema Poiska Avariynich Sudov," meaning "Space System for the Search of Vessels in Distress"

COTS Commercial Off-The-Shelf

CP Corrections Processor

CPDLC Controller Pilot Datalink Communications

CPR Cost Performance Report

CPS Chips per second

CPSD Carrier Phase Standard Deviation

CPU Central Processor Unit

CRC Cyclic Redundancy Check

CRE Central Region East (geo satellite East)

CRL Communications Research Laboratory

CRW Central Region West (geo satellite West)

CSC Computer Software Component

CSCI Computer Software Configuration Item

CSU Computer Software Unit

CV Correction & Verification

CW Continuous Wave

CWAAS Canadian WAAS

```
==D==
dΒ
     Decibel
dBi
     Decibel relative to isotropically radiated power
      Decibel relative to 1 Watt
dBW
DC
     Direct Current
DCAF
DCP
      Data Collection Processor (WRS)
DD
      Double Delta (in SQM tool)
DDD
      Database Driven Development
DFC Dole Frequency Contract
DGNSS Differential GNSS
DGPS Differential Global Positioning System
DH
      Decision Height
DLL
      Delay Loop-Lock (in HMI analysis)
DM
      Detection Metrics (in SQM ADD)
      Distance measurement equipment
DME
DNS
      Domain Name System
DNU
      Do Not Use (a SV setting indicating its data should not be used)
DOD
      Department of Defense (USA)
DOP
      Dilution of Precision
DQT
      Design Qualification Test
DRMS Distance Root Mean Square
DRS
      Data Reduction Software
DRT
      Deployment Readiness Test
D/T
      Detection Metric / Threshold
DT&E Development Test and Evaluation
==E==
EBC
      Event Based Certification
ECEF Earth-centered, earth-fixed (coordinates)
ECI Earth-centered inertial (coordinates)
ECR Engineering Change Request
EGNOS European (also Geostationary) Navigation Overlay System
EIRP Effective isotropic radiated power
EIRP
      Equivalent Isotropically Radiated Power
EMA
      Electromagnetic accelerator or accelerometer
EML
      Early Minus Late (in SQM tool)
ENU
      East-north-up (coordinates)
EOP
      Earth Orbit Parameters
      European Space Agency
      Electro Static Discharge<br/>
Streme Storm Detector
ESGN Electrically Supported Gyro Navigation (System USA)
ESS ESSMain Page | Enhanced Shadow System
EU
     European Union
EVM Earned Value Management
EWAN EGNOS Wide-Area (communication) Network (EGNOS)
==F==
FAA
      Federal Aviation Administration (USA)
FAR
      Federal Aviation Regulation - Part 171
FAS
      Final Approach Segment
FCS
     Frame Check Sequence (O&M access Router)
FCR
     File Change Report
FDI
     F D I (we need to find out)
FDP
     Field Deployment Plan
FEC
     Forward Error Correction
```

Fault Free (Fault Tree)

- FLL Frequency-lock loop
- FLP Full LPV Lateral Precision with Vertical guidance
- FM Frequency modulation
- FMEA Failure Modes and Effects Analysis
- FOB Freight-On-Board
- FOC Full Operational Capability
- FOG Fiberoptic gyroscope
- FPE Final prediction error (Akaike's)
- FRDF Facility Reference Data File
- FRP Federal Radionavigation Plan
- FSCK It is a Unix utility for checking and repairing file system
- FSLF Free-space loss factor
- FT Feet
- FTA Fault Tree Analysis
- FTD Fails To Detect (Fault Tree)
- FVS Functional Verification System

==G==

- GA General Aviation
- GAD Ground Accuracy Designator
- GAGAN GPS & GEO Augmented Navigation (India)
- GBAS Ground-based augmentation system
- GCCC Geo Code Carrier Coherence
- GCCS Geostationary Communication and Control Segment
- GCS Geostationary Communications Subsystem
- GCST Geostationary Communications Subsystem Type 1
- GDOP Geometric dilution of precision
- GEO Geostationary Earth Orbit

 Seosynchronous Satellite
- GES GPS Earth Station COMSAT

Ground Earth Station

- GFE Government-Furnished Equipment
- GIC GPS Integrity Channel
- GIM Global Ionospheric Model
- GIP Government Industry Partnership
- GIPSY GPS Infrared Positioning System
- GIS Geographic information system(s)
- GISE Grid Ionospheric Slant Error
 GIVE Grid Ionospheric Vertical Error
- GIVEI Grid Ionospheric Vertical Error Indicator
- GLONASS Global Orbiting Navigation Satellite System
- GLS Global Positioning System Landing System
- GNSS Global Navigation Satellite System
- GNSS Global Navigation Satellite System
- GNSSP Global Navigation Satellite System Panel
- GOA GIPSY/OASIS analysis
- GP GUS Processor
- GPS Global Positioning System
- GPSL GPS Leveling (GPSL) Tool | GSP Leveling HMI tool
- GPT GUS Processor Type 1
- GSP GEO Satellite Payload
- GSQA GEO Signal Quality Analysis
- GST GEO Satellite Transponder
- GUI Graphical User Interface
- GUS Geostationary Satellite Up-link Station
- GUSP GUS Processor
- GUST GEO Uplink Subsystem Type 1

==H==

HAC Height, Along, Cross

HAL Horizontal Alert Limit

HCL Height, Cross-track, and Along-track

hcp HMI Critical Parameters (HCP) | HMI Critical Parameters

HDOP Horizontal Dilution of Precision HFWG Human Factors Work Group

HMI Hazardously Misleading Information (HMI) | Hazardously Misleading Information

HNL Honolulu, WRS Site HOW Handover word HRB Hazard Review Board

HRG Hemispheric resonator gyroscope

HTEM HMI Tool Execution Manual (HTEM) | HMI Tool Execution Manual

HWCI Hardware Configuration Item

Hz Hertz

==|==

I-3 INMARSAT-3

I/S Interference-to-Signal ratio
IAD Incremental Asynchronous Data

ICAO International Civil Aviation Organization

ICAT

ICC Ionospheric correction computation

Inter-code comparison
Interface Control Contractor

ICD Interface Control Document

ICDLS Interim Contractor Depot Logistics Support

ICG International Committee on GNSS (United Nation)

ID Identifier

IDV Independent Data Verification (of WAAS)

IF Intermediate frequency

IFOG Integrating or interferometric Fiberoptic gyroscope

IFPR Ionospheric Free PseudoRange (HMI)

IGD Ionospheric grid Delay (HMI)

IGP Ionospheric grid point (for WAAS)

IGS International GNSS Service

ILS Instrument Landing System

ILS Integrated Logistic Support

IMC Instrument Meteorologic Condition

IM Integrity Monitor

IMS I M S

IMUT Utility Function (inside the Integrity Monitor)

IMVM Broadcast Message Validation IMU Inertial measurement unit

Inmarsat International Mobil (originally "Maritime") Satellite Organization

INS Inertial navigation system

IOC Initial Operating Capability

IOD Issue of Data

IODC Issue of data, clock

IODE Issue of data, ephemeris

IODF Issue of Data Fast Correction

IODI Issue of Data Ionosphere

IODP Ionosphere Issue of Data PRN

ION Institute of Navigation

IONO Ionosphere, Ionospheric

IOT In-orbit test

IPP Ionospheric Pierce Point

IR Implementation Report

IRB Inter-receiver Bias (in SQM, defined as the component of the detection metric measurements that are caused by receiver filter distortion)

- IRD Interface Requirements Documents
- IRS Interface Requirements Specification
- IRU Inertial reference unit
- ISA Inertial sensor assembly
- ISP Integrated Software Prototype
- ITRF International Terrestrial Reference Frame
- IUA Item Under Analysis

==J==

- JAA Joint Aviation Authorities (regulatory authorities of a number of European States, similar to FAA)
- JFS Journaled File System (is a 64-bit file system created by IBM)
- JPALS Joint precision approach and landing system for air force (similar to LAAS)
- JPL Jet Propulsion Laboratory
- JRC Joint Resource Council
- JTIDS Joint Tactical Information Distribution System

==K==

- Kbps Kilobits per second
- kHz Kilohertz
- km Kilometer
- KP KP is a index used to indicate the severity of the global magnetic disturbances in near Earth space.
- KPA Klystron Power Amplifier
- KTP Knowledge Transfer Plan
- KVM Keyboard Video Mouse

==L==

- L1 Radio Frequency L1 (GPS)
- L2 Radio Frequency L2 (GPS)
- LAAS Local Area Augmentation System
- LADGPS Local-area differential GPS
- LD Location determination
- LEM Lunar Excursion module
- LEO Low-Earth Orbiter
- LGF Local Area Augmentation System Ground Facility
- LHCP Left-hand circularly polarized
- LNA Low Noise Amplifier
- LNAV Lateral NAVigation (also referred to as NPA Non-Precision Approach)
- LNAV/VNAVLateral Navigation/Vertical Navigation (LNAV/VNAV)
- LORAN Long-range navigation
- LOF Loss of Function (HMI)
- LOS Line of sight (measurement in HMI)
- LPV Localizer Performance with Vertical Guidance (preferred), "'OR"' Lateral Precision with Vertical Guidance "'OR" Lateral positioning with vertical guidance
- LRU Line Replaceable Units
- LSB Least Significant Bit
- LSP Local Status Panel
- LTC Linear Time Code
- LTC Longitudinal Time Code
- LTI Linear Time-Invariant (in Kalman Filter)
- LTN Littleton, CO
- LTP Local Area Augmentation System Test Prototype
- LTP Local tangent plane

==M==

M Meter

MASPS Minimum Aviation System Performance Standard

MBOC Modified BOC

Mbps Million bits per second

M&C Monitor & Control

MCC Mission/Master Control Center (EGNOS)

Mchips Megachips

MCP Message Control Processor
MCPS Million Chips Per Second
MCS Master Control Station

MDE Minimum Detectable Error (in CCC tool)

MDT Maintenance Data Terminal

MEDLL Multipath-estimating delay-lock loop MEMS Microelectromechanical system(s)

MEO Medium Earth Orbit

MERR Maximum Error Range Residual

MHz Megahertz

MI Misleading Information (Fault Tree)

MIL-STD Military Standard

MIPS Master Integrated Program Schedule

ML Maximum likelihood

MLA Multipath Limiting Antenna

MLE Maximum likelihood estimate (or estimator)
MMSE Minimum mean-squared error (estimator)

MMT Multipath mitigation technology

MOPS Minimum Operational Performance Standards
MPR Porta Vallarta Reference Station in Mexico

MPS Maintenance Processor System

ms millisecond

MSAS MTSAT Satellite-based Augmentation System (Japan)

MSB Most Significant Bit MSL Mean Sea Level

MTBF Mean Time Between Failure

MTL Multi-Tool Launcher

MTSAT Multifunction Transport Satellite (Japan)

MTTR Mean Time to Repair

MVUE Minimum-variance unbiased estimator mW-hr/cm2Milliwatt-hours per centimeter squared

mW/cm2 Milliwatts per centimeter squared

MWG Momentum wheel gyroscope

==N==

NA Not Applicable

NANU Notice Advisory to NAVSTAR Users

NAS National Airspace System

NAVSTAR Navigation system with time and range

NAWC Naval Air Warfare Center

NCO Numerically controlled oscillator

NCP NAS Change Proposal

NDI Non-Developmental Item

NDU Navigation Data Unit

NED North-east-down (coordinates)

NFF Non Fault Free (Fault Tree)

NGS National Geodetic Survey (USA)

NICS NAS Interfacility Communications System

NIST National Institute of Standards and Technology

NLES Navigation Land Earth Station(s) (EGNOS)

ns nanosecond

NOTAM Notice to Airmen

NPA Nonprecision Approach (see also LNAV - Lateral Navigation)

NSE Navigation Sensor Error

NSRS National Spatial Reference System

NSTB National Satellite Test Bed NTP **Network Time Protocol** ==0== .odiff Formatted Orbit Differential File O&M Operations and Maintenance Station OAC O&M Access Router OASIS Orbit analysis simulation software OBAD Old but active data OCC Operational Control Center OCXO Oven Controlled Crystal Oscillator Orbit determination OEI O & M External Interface OLM Off Line Monitoring OPUS Online Positioning User Service (of NGS) Open service (of Galileo) Operational System Parameter **OSP** ==P== P-Code Precise or Protected Code P3I Pre-Planned Product Improvement Precision Approach PACF Performance Assessment and Checkout Facility (EGNOS) PAN Performance Analysis Network (part of WAAS Performance Analysis Report) PAT&E Production Acceptance Test and Evaluation PBF Position Bias Filter PBLI Product Baseline Listing Index PBN performance Based Navigation Problem/Change Report (PNB.1.2.2.2, p.35) "'or'" Problem Change Request (PNB.1.2.2.2, p.12) [Note multiple definitions, even in the same document. Probably all mean the same thing] PCRB Program Change Review Board **PCS** PDF Portable Document Format

Probability Distribution Function PDOP Position dilution of precision PDL Program Design Language **Program Execution Plan** PHMI Probability Hazardously Misleading Information Proportional and integral (controller) PID Process Input Data (of WAAS) PID Proportional, integral, and differential (control) PIGA Pulse integrating gyroscopic accelerometer PLL Phase-lock loop PLCI unknown. see PNB.1.2.2.4, p.6 PLRS Position Location and Reporting System (U.S. Army) PN **PRN Number** PNB (Raytheon) Program Notebook PNE Phase Noise Enhancer PNT **Positioning Navigation and Timing** POM Program Overview Meeting POR Pacific Ocean Region PPD **Personal Privacy Devices** PPE Personal Protective Equipment PPFL Pierce Point Filter Line PPS **Precise Positioning Service** PPS Pulse(s) per second PR Pseudorange PRCf Pseudorange Correction fast

PROTCV Prototype Correction and Verification

```
Public Regulated service (of Galileo)
PRSD Pseudo Range Standard Deviation
PSD
      Power spectral density
PUID Project Unique Identifier
PVR
      Performance Validation Record (HMI)
==Q==
QΑ
      Quality Assurance
==R==
RAG
      Receiver antenna gain (relative to isotropic)
RAIM
                                                   Autonomous
                                                                     Integrity
                                                                                   Monitoring
                                                                                                              RAIM
                                                                                                                         in
                                     Receiver
                                                                                                    (see
Wikipediahttp//en.wikipedia.org/wiki/Receiver_Autonomous_Integrity_Monitoring)
       Random-Access Memory
RAM
RCM
       Reliability Centered Maintenance
RDM
       Range Domain Monitor
      Relative Detection Time (in SQM ADD)
RDT
      Raytheon Engineering Authority "'aka" Lead Engineer (PNB.1.2.2.2, p.6) "'OR" Responsible Engineering Activity (PNB.1.2.2.4, p.1)
REA
RF
     Radiofrequency
RFU
      Radio Frequency Uplink
RG
     Rate Group
RGC
      Range Group Converter
RHCP Right-hand circularly polarized
RIMS
      Ranging and Integrity Monitoring Stations(s) (EGNOS)
RINEX Receiver Independent Exchange format (for GPS data)
RLG
      Ring laser gyroscope
       Reliability, maintainability, availability
RMA
RMMS Remote Maintenance Monitoring System
RMS/rms root mean square
RMS
      Reference monitoring station
RMS
      Remote Monitoring Subsystem
RNP
      Required Navigation Performance
      ROT
ROUS Rodents Of Unusual Size
RPY
      Roll-pitch-yaw (coordinates)
RR
     Reference Receiver
RS
     Reference Station
RSP
     Remote Status Panel
RSS
      Root Sum Square
RTC
      Real Time Control
RTCA Radio Technical Commission for Aeronautics, Inc., Washington, DC
      Real Time Correction Messages
RTCM Radio Technical Commission for Maritime Service
RTOS Real-time operating system
RVCG Rotational vibratory coriolis gyroscope
RVR
      Runway Visual Range
==S==
    second
     Selective availability (also abbreviated "S/A")
SA
SACIL Safety Assurance Configuration Index List
      Search and rescue (service of Galileo) "'OR" Safety Assessment Report
SARP Standards and Recommended Practices (Japan)
SARPS Standards and Recommended Practices
SARSAT Search and rescue satellite-aided tracking
SAS Software Accomplishment Summary
SATCOM or Satcom which stands for Satellite Communication
SAW Surface acoustic wave
```

PRN

PRS

Pseudorandom noise (see also PN and SVN)

```
SBIA Safety Baseline Impact Assessment
SBIRLEO Space-based infrared low earth orbit
SC Special Committee
SCAF A Receiver Error (Scale Cross Ambiguity Function)
SCC Satellite Control Center
SCCB Software Change Control Board "OR" Software Configuration Control Board
SCM
      Software Configuration Management
SCN
SCOUT Scripps coordinate update tools
      Satellite Correction Processing (of WAAS)
      Specification Change Request (PNB.1.2.2.2, p.36) "'or'" (aka Specification Change Report but probably not official) "'or'" Software
Change Request (PNB.1.2.2.4, p.2 PNB.1.2.2.2, p.12) [Probably incorrect]
     Signal Deformation
SDF
      Software Development Folder
SDR
     Software Design Review
     System Design Review
SEWG Systems Engineering Working Group
SF
   Scale Factor
SGS Signal Generator Subsystem
SGST Signal Generator Subsystem Type 1
SIS Signal in space
SIT System Integration Test
SLB Short Loop Back (Fault Tree)
SLES Second Level Engineering Support
SLOC Single Line of Code (term often used for budgeting purposes in level B code)
      Solar magnetic
     Sigma Manager (SM) in HMI Tool | Sigma Manager in HMI tool
     Safety Monitor(SM) in C&V | Safety Monitor in C&V
SMS
      Safety Management System
SMS
      Service Monitoring Subsystem
SNAS Satellite Navigation Augmentation System (China)
SNMP Simple Network Management Protocol
SNR Signal-to-noise ratio
SOL
      Safety of Life Service (of Galileo) <b>OR </b> S*** Out Of Luck
SOS
      Safety and Operation Support
SOW
      Statement of Work
SΡ
     Safety Processor
SPS
     Software Product Specification (C&V)
     Standard Positioning Service (GPS)
SQM
       Signal Quality Monitor/Signal Quality Monitoring
SRM
      Safety Risk Management
SRR Software Release Record
SRS
     Software Requirements Specification
SSD System Support Documentation
SSM System Support Modification
SSP
     Site Specific Parameters
SSWG System Safety Working Group
STD Software Test Descriptions
     Signal Task Force (of Galileo) <b>OR </b> Software Test Facility
STF
STP
     Software Test Plans
     Space Vehicle
SV
SVM Service Volume Model
      Space Vehicle Number (=PRN for GPS)
SVN
==T==
TACAS Traffic Alert and Collision Avoidance System
TBB To Be Broadcast (In HMI GIVE section GIVE_TBB)
```

SBAS Space-based augmentation system

TBD To Be Determined

TCAS Traffic Collision Avoidance System

TCN Terrestrial Communications Network

TCS Terrestrial Communications Subsystem (for WAAS)

TCXO Temperature-compensated Xtal (crystal) oscillator

TDMA Time Duplex Multiple Access

TDOA Time difference of arrival

TDOP Time dilution of precision

TDS Time Distribution System

TEC Total Electron Content

TECU Total Electron Content Units

TIB WAAS Technical Instruction Book | Technical Instruction Book

TLM Telemetry word

TLT Test Loop Translator

TM Threat Model Tool

TO Technical Operation

TOA Time of Applicability

TOA Time of arrival

TOD Time of Day

TOF Time of Applicability (is this correct? see TOA above)

TOW Time of Week

TRACON Terminal Radar Approach Control

TSO Technical Standard Order

TSS Test Support Software

TSWM Time Stamped WAAS Message (Fault Tree)

TTA Time to Alarm or Time to Alert

TTFF Time to first fix

TTT Time-to-Transmit (in SQM ADD)

TWG Transition Working Group

==U==

U.S. United States

UB UIVE Bounding Tool

UDRE User Differential Range Error

UDREI User Differential Range Error Indicator (SRS, p.C-612) "'OR'" UDRE Index (see HMI Doc A014_011 p220) where indices range from 3 to 13. Each UDREI represents a specific UDRE.

UE User Equipment (i.e. the WAAS-enabled GPS reciever a typical user would be operating to obtain WAAS corrections. See MOPS DO-229D)

UERE User-equivalent range error

UGP User Grid Points

UIVE User Ionospheric Vertical Error

UL Underwriter Laboratories

UPM User Position Monitor (in Safety Processor) or User Position Monitor Tool

URA User Range Accuracy

URE User range error

USAF United States Air Force

USN United States Navy

USNO US Naval Observatory

USTS

UTC Universal Time, Coordinated (or Coordinated Universal Time)

UTM Universal Transverse Mercator

==V==

VAG Visitor Access Gateway

VAL Vertical alert limit

VCG Vibratory coriolis gyroscope

VDB Very High Frequency Data Broadcast

VDD Version Description Document

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VDOP Vertical Dilution of Precision
VHF
      Very high frequency (30-300 MHz)
VME
      Versa Module Eurocard
VNAV Vertical Navigation
VOR VHF Omnirange (radionavigation aid)
VPE
      Vertical Position Error
VPI
     Vertical Protection Levels
VRTM Verification Requirements Traceability Matrix < OR> Verification Requirements Test Matrix
VRW Velocity Random Walk
==W==
W3P
      WAAS 3rd Generation Prototype
W3CP WAAS 3rd Generation Correction Processor Prototype
W3SP WAAS 3rd Generation Safety Processor Prototype
WAAS Wide Area Augmentation System
WADGPS Wide-area differential GPS
      Woodbine, Maryland
WCCB WAAS Change Control Board
WCN
      WAAS Communications Node
WCR WAAS Change Request
WEI WAAS External Interface
WFO WAAS Follow On {contract}
WG
      Working Group
WGS
      World Geodetic System
WIPP
      WAAS Integrity & Performance Panel
       Wide Area Augmentation System Integrigy & Performance Panel Integrity Resolution Process
WIRP
       WAAS Integrated Software Prototype (equivalent to PROT_SP) (check on this definition) <b>OR</b> WAAS Integrity Simulated
WISP
Safety Processor
WJHTC William J. Hughes Technical Center - Atlantic City, NJ
WLF
      WAAS Lab Facility
WMP
       WAAS Message Processor
       Wide Area Master Station | Wide-area Master Station
WMS
WN
      Week Number
WNT
      WAAS Network Time
WOS
      WAAS Operations Specialist
WPR
       WAAS Problem Report
WRE
       Wide-area Reference Equipment (WRE) | Wide-area Reference Equipment
WRS
      Wide Area Reference Station | Wide-area Reference Station
WRT
      With Respect To
WTMS WAAS Training and Maintenance System
WUM
       WAAS Users Message <b>OR</b> WAAS User Message
WUPS WAAS Users Position Solution <b>OR</b> WAAS User Position Software
==X==
==Y==
==Z==
      Albuquerque Air Route Traffic Control Center (ARTCC)
ZAB
      Chicago Air Route Traffic Control Center (ARTCC)
ZBW
      Boston Air Route Traffic Control Center (ARTCC)
      Washington Air Route Traffic Control Center (ARTCC)
ZDC
ZFTZ
      Zero Filled Type Zero
ZFW
      Dallas-Ft Worth Air Route Traffic Control Center (ARTCC)
ZHU
      Houston Air Route Traffic Control Center (ARTCC)
      Indianapolis Air Route Traffic Control Center (ARTCC)
ZJX
     Jacksonville Air Route Traffic Control Center (ARTCC)
ZKC
      Kansas City Air Route Traffic Control Center (ARTCC)
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ZLA

ZLC

Los Angeles Air Route Traffic Control Center (ARTCC)

Salt Lake City Air Route Traffic Control Center (ARTCC)

- ZLG Zero-Lock Gyroscope ("Zero Lock Gyro" and "ZLG" are trademarks of Northrop Grumman Corp.)
- ZMA Miami Air Route Traffic Control Center (ARTCC)
- ZME Memphis Air Route Traffic Control Center (ARTCC)
- ZMP Minneapolis Air Route Traffic Control Center (ARTCC)
- ZNY New York Air Route Traffic Control Center (ARTCC)
- ZOA Oakland Air Route Traffic Control Center (ARTCC)
- ZOB Cleveland Air Route Traffic Control Center (ARTCC)
- ZSE Seattle Air Route Traffic Control Center (ARTCC)
- ZSU San Juan Air Route Traffic Control Center (ARTCC)
- ZTL Atlanta Air Route Traffic Control Center (ARTCC)